



# VIBRATIONS ANALYSIS OF LAMINATED COMPOSITE AND EULER BERNOULLI BEAMS USING DIFFERENTIAL TRANSFORMATION METHOD

By

#### Eng. SOUMA MOHAMED ABDEL-GHANI MOHAMED

A Thesis Submitted to the Faculty of Engineering at Cairo University In Partial Fulfillment of the Requirements for the Degree of

**DOCTOR OF PHILOSOPHY** 

IN

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Under the supervision of

Prof. Dr. MOHAMED. M. NASSAR

Professor of Engineering Mechanics Dept. of Eng. Math. and Physics Faculty of Eng. Cairo University Prof. Dr. ADEL A. ABDEL GAWAD

Professor of Engineering Mechanics
Dept. of Eng. Math. and Physics
Faculty of Eng. Cairo University

Dr. KAREM. M. EWIS

Lecturer of Engineering Mathematics Dept. of Eng. Math. and Physics Faculty of Eng. Fayoum University

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Approved by the

**Examining Committee:** 

Prof. Dr. MOHAMED. M. NASSAR

(Thesis Main Advisor)

Prof. Dr. ADEL A. ABDEL GAWAD

(Member)

Prof. Dr. ALI ABDEL-AZEM HASSAN

(Internal Examiner)

Prof. Dr. ELSAYED SAAD ABDEL-SALAM

(External Examiner)

Professor, Engineering, Zagazig

Engineer: souma Mohamed Abdel-Ghani Mohamed

Date of Birth: 9/ 1 / 1982
Nationality: Egyptian

E-mail: sma01@fayoum.edu.eg

Phone.: 01140666453
Address: El- Fayoum
Registration Date: 1/10 / 2010

Awarding Date: / /

Degree: Doctor of Philosophy

**Department:** Engineering Mathematics and physics

**Supervisors:** Prof. Dr. Mohamed Mohamed Nassar

Prof. Dr. Adel Ahmed Abdel Gawad

Dr. Karem Mahmoud Ewis (Lecturer- Engineering Fayoum)

Examiners: Prof. Dr. Elsayed Saad Abdel-Salam (External Examiner)

(Professor-Engineering – Zagazig)

Prof. Dr. Ali Abdel-Azem Hassan (Internal Examiner)
Prof. Dr. Mohamed Mohamed Nssar(Thesis Main Advisor)

Prof. Dr. Adel Ahmed Abdel Gawad (Member)

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Key Words: Vibrations; Laminated beams; Composite beams; Euler Bernoulli beams; Differential Transformation Method

#### **Summary:**

Differential Transformation Method is a technique that uses Taylor series for the solution of differential equations in the form of a polynomial. It is different from the high-order Taylor series method, which requires symbolic computation of the necessary derivatives of the data functions. The Differential Transformation Method leads to an iterative procedure for obtaining an analytic series solutions of functional equations. Beams with abrupt changes of cross-section are used widely in engineering. Rectangular and circular shapes of the cross section are chosen because it often occurs in practice. In this thesis, the vibration problem of Euler-Bernoulli beams has been solved using Differential Transformation Method. The beam has variable cross sections and various end conditions.



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## NOMENCLATURE

Symbol	unit	Identification
E	GPa	
I	$m^4$	Inertia of the beam
v	M	Vertical displacement of the beam
Å	$m^2$	
h	M	
b	M	Width of the beam
L	M	
$\overline{ ho}$	$Kg/m^3$	Density of the beam material
T	J	Kinetic energy
V	J	Strain energy
$\omega$	Hz	Beam natural frequency
arOmega	-	
$\overline{k}_1$	$N/m^2$	Linear parameter of the foundation
$\frac{\overline{k}}{k_3}$	$N/m^4$	Non-Linear parameter of the foundation
$\overline{\mu}$	$N S/m^2$	Damping parameter of the foundation
$\delta$	-	Dirac delta function
$\overline{\mathcal{Q}}$	KN	
$\overline{v}$	m/s	
t		Time
$T_n$	S	
$M_{\scriptscriptstyle X}$	N m	Bending moment
$M_{xy}$	N m	Torque
$N_x$	N	
$Q_{xz}$	N	
$k_s$	-	Shear correction factor
$oldsymbol{ heta}$	degree	Angle betw. fiber directions and longitudinal axis of laminated beam
$v_{12}$	_	
$G_{12}$	GPa	Shear modulouls of elasticity
β	-	Coefficient of variation of beam cross section
N	-	Terms number
M	-	Terms number
k	-	Index
$E_I$	GPa	Young's modulus of the laminated beam material in the
	CD	fiber direction.
$E_2$	GPa	Young's modulus of the laminated beam material in the perpendicular on the fiber direction.
$Q_{ij}$	-	The transformed reduced stiffness constants of laminated beam
$\stackrel{\mathcal{L}y}{A_{ij}}$	-	
$B_{ij}^{'j}$	-	The coupling stiffness constants of laminated beam
$\overset{-}{D}_{ij}$	-	The bending stiffness constants of laminated beam